SHELE GOTT ONLY

SID 62-99-10

MONTHLY WEIGHT AND BALANCE REPORT

FOR THE APOLLO SPACECRAFT

CONTRACT NAS 9-150

1 December 1962

4.5.4.5

ASPO FISTRIBUTION COPY Destroy when no longer in use Do not return to ASPO file

Prepared by

Weight Control Group

CLASSIFICATION CHANGE

By authority of Document Master Control Station, NASA Changed by Document Master Control Station Formation Scientific and Technical Information Formation Scientific and Technical Information Scientific and Technical Information Factors

NORTH AMERICAN AVIATION, INC. SPACE and INFORMATION SYSTEMS DIVISION

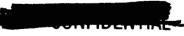




TABLE OF CONTENTS

	ITEM	PAGE
ı.	INTRODUCTION	1
II.	MISSION WEIGHT, CENTER OF GRAVITY AND INERTIA SUMMARY Apollo LOR Mission Apollo EOR Mission Command Module Weight and Center of Gravity Summary	2 3 4
III.	CURRENT WEIGHT STATUS Spacecraft Weight Status Summary Command Module Weight Status Command Module Changes Service Module Weight Status Service Module Changes Launch Escape System Weight Status Launch Escape System Weight Changes Adapter Weight Status	5 6 7-13 14 15-17 18 19
IV.	WEIGHT HISTORY	21-24
٧.	POTENTIAL WEIGHT AND C.G. CHANGES	25 –2 6
VI.	SPACECRAFT DETIIL WEIGHT STATEMENT	27 -49



INTRODUCTION

In this report the authorized weight column has been removed from the module status sheets. A revised format for the EOR and LOR mission summary is included indicating centers of gravity and inertias incorporating estimated values for the LEM. A more detailed breakdown is shown for the S/M propulsion system.

The major changes in the command module weight since the last report are in the structure, guidance and navigation, and electrical systems. The current basic structural weight is based on drawings which are about 82% calculated. The 40% calculated value shown on the status summary is lower because of the ablator weight which is assumed to be estimated. Potential weight reductions in structure are discussed in the potential weight changes but cannot be reflected in the current weight until they are accomplished by released drawing change.

The major changes in the service module weight since the last report are in the structure and electrical system. The current basic structural weight is based on a revised estimate of the structural design and of the insulation. The potential weight reduction reflected in the November report was accomplished and incorporated. In addition, the structural insulation weight was reduced about 200 pounds by utilizing a lower density material. The weight increase in the electrical system is due to incorporating increases in the P & W and Beech status reports.

The total spacecraft weight at injection increased by 70 lbs. and at S/M burnout 30 pounds. The current injected weight with the service module loaded with sufficient propellant to provide a 10 per cent Δ V margin is 84,465. This is based on LEM weight, including crew, of 25,000 lb.

APOLIO LOR MISSION

WEIGHT, CENTER OF GRAVITY AND INERTIA SUMMARY

VGMT	WEIGHT	CENTER	CENTER OF GRAVITY*	ITY*	MOMENTS OF	MOMENTS OF INERTIA (SIUG-FT ²)*	UG-FT ²)*
WITT	POUNDS	Х	Y	2	ROLL (X)	PITCH (Y)	XAW (Z)
COMMAND MCDULE	9350	1042.6	C7 FF	8.2	3934	3989	3774
SERVICE MCDULE - Less Propellant	10205	908.5	7.0-	-0.9	7509	12576	12413
TOTAL - Less Propellant	19555	972.6	7.0	3.5	11536	35602	35078
PROPELLANT	37190	905.8	5.6	-2.4	19300	20200	26100
TOTAL - With Propellant	26745	928.8	3.8	7.0-	31363	68253	73611
LAUNCH ESCAPE	6655	1295.2	0	0	228	8679	8879
LUNAR EXCURSION MODULE	54460	623.0	0	0	13616	12776	13247
ADAPTOR - LEM	3260	645.0	0	0	6979	z69†1	7697
TOTAL - SPACECRAFT LAUNCH	02116	863.3	2.4	7.0-	51381	153711	759605

*Centers of gravity are in the MASA reference system except that the longitudinal axis has an origin 1000 inches below the pangency point of the command module substructure mold line. NOTE:

APOLLO EOR MISSION

WEIGHT, CENTER OF GRAVITY AND INERTIA SUMMARY

	WEIGHT	CENTER	CENTER OF GRAVITY*	ITY*	MOMENTS C	MOMENTS OF INERTIA (SIUG-FT ²)	SLUG-FT ²)
ITEM	POUNDS	Х	Y	2	ROLL (X)	PITCH (Y)	YAW (Z)
COMMAND MODULE	9350	1042.6	1.2	8.2	3934	3989	3774
SERVICE MODULE - Less Propellant	10205	908.5	7.0-	-0.9	7509	12576	12413
TOTAL - Less Propellant	19555	972.6	7.0	3.5	11536	35602	35078
PROPELLANT	17875	900.2	27.0	-n.7	0019	7500	8200
TOTAL - With Propellant	37430	938.0	13.1	-3.8	19529	54138	55360
LAUNCH ESCAPE	9999	1295.2	0	0	228	8879	8879
ADAPTER - C-1	800	770.0	0	0	831	295	562
TOTAL - SPACECRAFT LAUNCH	538771	0.886	10.9	-3.2	20836	227707	22 9 06 1

*Centers of gravity are in the NASA reference system except that the longitudinal axis has an origin 1000 inches below the tangency point of the command module substructure mold line. NOTE:

COMMAND MODULE

WEIGHT AND CENTER OF GRAVITY SUMMARY

	VEHICLE CONFIGURATION	WT.	LAUNCH ABORT CONDITION	r CONDITI	NO		ENTRY CONDITION (LUNAR MISSION)	NDITION ISSION)	
		WEIGHT	X	Y	2	WEIGHT	I	I	2
	Earth Launch	(6320)	1042.6	1.2	8,2	(0\$66)	1042.6	1.2	8,2
	Add: Unexpended Waste & Water	1	1	1	ı	211	1	1	ı
	Shift Grew to Entry Position	1	1	ı					
						(0/63)	3 6/01	7	0
	Frior to Entry	1	I	l	•	(20%)	70446.7	• 1	7.7
4	Less: Propellant RCS	ı	1	1	ı	-210	1022.6	3.5	52.0
	Ablation Material Burnoff	1	1	ı	1	-314**	1019.7	0	11.2
	Nose Cone & Discone Antenna	-287*	1100.8	0.0	0.0	-264	1099.2	0.0	0.0
	Drogue Chute	-25	1090•0	11.0	-22.0	-25	1090.0	11.0	-22.0
	Prior to Main Chute Deployment	(9038)	1040.6	1,2	8.5	(6798)	1041.9	9.0	9.2
	Less: Main Parachutes (3)	-385	1092.0	0.0	9. 8	-385	1092.0	0.0	8.6
	Shift Crew to Landing Position								
	Landing	(8653)	1038.3	1.3	8.6	(8564)	1039.6	0.7	₽ •
SID 62-99-10	*Represents nose cone with ablative mater **Based on 21% ablation burnoff.	terial intact.	et.						
·/ (



SPACECRAFT

WEIGHT STATUS SUMMARY

	PREVIOUS STATUS	CHANGE TO	CURRENT WEIGHT	BASIS	FOR CUF	RENT
ITEM	11-1-62	CURRENT	12-1-62	%EST	%CAL	%ACT
COMMAND MODULE	8760	+590	9350	68	31	1
SERVICE MODULE	55765	-560	55205 °	14	86	
LES	6495	+160	6655	42	58	
ADAPTER	3260		3260 -	100		
TOTAL	74280	+190	74470	27	73	

COMMAND MODULE WEIGHT STATUS

NGEL	PREVIOUS	CHANGE	CURRENT	BASIS	S FOR C	FOR CURRENT
	11-1-62	CURRENT	12-1-62	%EST.	%CAL.	%ACT.
Structure	6707	£ \$ 7+	74782	\$	2	
Crew Systems	520	2-	513	700		
Communication and Instrumentation	106	-12	688	700		
Guidance and Navigation	357	+74	731	30 0		
Stabilization and Control	201	+13	214	100		
Reaction Control	250	+17	267	100		
Electrical Power	385	777+	627	100		
Environmental Control	253	6+	262	100		
Earth Landing	502	6-	667	82	55	25
WEIGHT EMPTY	7398	+562	0962	728	56	~
Crew (3) (50, 70, 90 Percentile)	528		528		90	
Suits and Personal Equipment	132	+5	134	100		
Survival Water	18		18		100	
Food and Containers	8		8	100		
Reaction Control Propellant	210		210 ~		100	
Environmental Control Chemicals	134	¥	140		100	*
Scientific Payload	250		250′	001		
GROSS WEIGHT	8760	+590	93 5 0	67	32	п

CONFIDENTIAL



COMMAND MODULE

CURRENT WEIGHT EMPTY CHANGES

STRUCTURE (+	+4	5	3	

The current Command Module structural weight of 4482 pounds is based on a high percentage of released and pre-released drawings for the primary inner structure and heat shield structure. Approximately \$2% of the primary structural weight has been calculated based on these two types of drawings.

Due to the current weight increases, an intensive weight reduction program has been initiated to review the criteria and design. The preliminary results have indicated several areas where potential significant weight changes exist. These potential weight reductions are in the following categories:

- 1. Removal of weight penalties incorporated into present design due to impact attenuation provisions.
- 2. Reduction of ablator to be consistent with weights being negotiated with Avco.
- 3. Changes to existing design concepts to optimize design, improve load paths and increase structural efficiency.

The potential weight reductions listed above are discussed in more detail in Section V (Potential Weight and C.G. Changes).

CREW SYSTEMS	(-7)
Add personal radiation dosimeter for three crewman per NASA requirement.	+5
Transfer Bio-medical instruments to useful load.	- 5
Reduce personnel communication system per revised estimate.	-4
Reduce sleeping restraints per revised estimate.	-4
Delete sleeping and privacy curtains per latest requirement evaluation.	- 2
Reduce waste management per calculations in lieu of estimates.	- 5
Increase lighting system per revised estimate based on current requirements.	+11



COMMAND MODULE

CURRENT WEIGHT EMPTY CHANGES

CREW SYSTEMS (Continued)	
Increase survival kit per revised estimate	+1
Increase food reconstitution device per revised estimate	+2
Delete extra vehicular accessories and add hatch egress accomodations	- 9
Increase lap board per revised estimate	+1
Decrease in-flight test vest requirement to provide for tool belts only	-1
Add food preparation table per revised requirements	+3
COMMUNICATIONS AND INSTRUMENTATION	(-12)
Communications	
Reduce DSIF power amplifier due to revised vendor estimate	- 3
Increase VHF/2-KMC omni antenna to incorporate different radome material and different foam potting, due to revised temperature requirements	+14
Decrease antennas and transmission lines due to reduced estimate for HF recovery antenna	-1
Decrease instrumentation sensors due to revised estimate	-1 5
Controls and Displays	
Addition, deletion and relocation of various displays and controls at the Control Station, Center Station, System Management Station, RH console and LH console to simplify controls	- 9
Add:	
Service Module Quad. Temp. Ind. +3 SCS Power Control Panel +2 IFTS Scan Select Panel +1 Lighting Control (2) +2 Page t France and Forth Londing +3	



COMMAND MODULE

CURRENT WEIGHT EMPTY CHANGES

COMMUNICATION AND INSTRUMENTATION (Continued)

Delete:

Docking Controls	- 2
Rate of Climb	-1
Landing Display Control	- 2
Earth Landing System	-4
Launch Escape System	-4
Egress Hatch Jettisoning System	- 2

Revise:

Integrated Display	+2
SCS Control Panel	- 2
Reaction Control System	-2
Service Module Propulsion	-3

Add loudspeaker at	NASA request defined in abstract of	
electrical system	meeting number 9 per letter	
11120 MA	• • •	

GUIDANCE AND NAVIGATION

(+74)

+2

Change the weight of the followin	g items in accordance with	
the latest MIT Report:	+	164

Sextant	- 15
Telescope	+1
Mavigation Base	-1
Power Servo Assy	+5
Cabling	+34
Spares	+40

Increase e	stimated NAA	cabling	+10

STABILIZATION	AND	CONTROL	(+13

Add BMAG Coupler	+13
------------------	-----

REACTION CONTROL SYSTEM (+17)

Increase t	he following items due to relocation of the tanks
to the +	Z axis to improve the reentry center of gravity.
The incr	ease is due to longer piping runs and larger tankage
to store	the increased unusable propellants.

+20



COMMAND MODULE

CURRENT WEIGHT EMPTY CHANGES

REACTION	CONTROL	SYSTEM	$({ t Continued})$
----------	---------	--------	--------------------

Propellant System

Tanks +2
Plumbing +13

Pressure System

Tanks +4
Plumbing +1

Change the following items based on calculations of revised schematics:

-3

Propellant System

Vа

Valves and Regulators +2 Sensors -3

Pressure System

Valves and Regulators +2 Sensors -4

ELECTRICAL POWER

(+44)

Increase installation provisions for the energy souce, power conversion and Electrical Utility System per revised estimates

+2

Increase electrical utility umbilicals per calculations from layout drawings and include the transfer of ablative material from ablator and substructure.

142

ENVIRONMENTAL CONTROL SYSTEM

(+9)

Change the following components per negotiations with Airesearch to further define the "C" change which removed certain redundancies and was part of a cost reduction:

-41

Pressure Suit Components

Subcontracter Components -8
Ducting, Conn., Clamps, etc. -2





COMMAND MODULE

CURRENT WEIGHT EMPTY CHANGES

ENVIRONMENTAL CONTROL	SYSTEM	(Continued)	
-----------------------	--------	-------------	--

Water Glycol Circuit		
Subcontractor Components Water Glycol Plumbing, etc.	+5 -3 -11	
Pressure and Temperature Control		
Subcontractor Components Ducting	-1 -4	
Oxygen Supply		
Subcontractor Components Plumbing	+1 -2	
Water Supply		
Subcontractor Components Plumbing	-2 -4	
Supports	-10	
Add the following items per Airesearch	n Weight Report:	+33
Brackets, Plumbing, Electrical Wiring	+20	
Instrumentation	+13	
Add gas analyzer, which is NASA furnis	shed	+17
EARTH LANDING		(-9
Increase the following items per North	nrop Status Report:	+10
Main Cluster Sequence Controller	+6 +4	, 20
Reduce location aids due to removal of	f the Shark Replellant	
Ejector	•	-1
Reduce attach provisions per calculati	ions versus estimates	_7



COMMAND MODULE

CURRENT WEIGHT EMPTY CHANGES

EARTH LANDING (Continued)

Reduce Forward Heat Shield Release System for calculations versus estimates	-6
Reduce electrical pyrotechnic initiation provisions based on re-evaluation of system requirements	- 5
OTAL COMMAND MODULE WEIGHT EMPTY CHANGES	+582



COMMAND MODULE

CURRENT USEFUL LOAD CHANGES

Transfer biomedical instruments from Crew Systems weight empty (+5 pounds) and re-estimated weight required to be 2 pounds in lieu of 5 pounds.	+2
Reduce personal hygiene equipment per re-evaluation of requirements and weight of personal shaver, ingestible dentifrice package, fecal and emesis bag set.	- 3
Increase medical equipment due to reanalysis of requirements for 14 day mission.	+1
Increase chemical disinfectant per revised estimate	+2
Add containers for LiOH and Charcoal	+6
TOTAL COMMAND MODULE USEFUL LOAD CHANGES	+8



SERVICE MODULE WEIGHT STATUS

NOGEL	PREVIOUS	CHANGE	CURRENT	BASIS	FOR CURRENT	RRENT
MALL.	31A103 11-1-62	CURRENT	ME1GH1 12-1-62	%EST	&CAL	%ACT
Structure	3317	779-	2673	100		
Electronics	172	9	166	300		_
Reaction Control	179	-50	591	100		
Electrical Power	1601	+146	1243	100		
Environmental Control	98	-28	58	100		
Propulsion System Engine Installation Propellant System	606 2465	6-	606 2456	100		
WEIGHT EMPTY	8384	-591	7793	100		
Usable RCS Propellant	7611		611	•	100	
Usable Supercritical Reactants	¥ 1627	27-	127		100	
Environmental Control Fluids	191	+11	202		700	
Main Propulsion Helium	139		139		100	
Main Propellant Residuals Trapped Mixture Ratio Tolerance Loading Tolerance	(900) 225 450 225		(900) 225 450 225		100	
Unusable RCS Propellant	19		61		100	
Unusable Supercritical Reactions	√9	+62	89		100	
BURNOUT WEIGHT	10765 .	-560	10205	9/	57	
Main Propellant	45000		45000		100	
GROSS WEIGHT	55765	-560	55205	17	98	

-CONFIDENTIAL SID 62-99-10



SERVICE MODULE

CURRENT MEIGHT EMPTY CHANGES

STRUCTURE	(-644)
Decrease the radial beams per load and structural configuration requirements	-200
Decrease the internal structure and engine compartment closeout per revised estimate and structural configuration	-88
Decrease the outer shell skin gage and honeycomb core depth per load requirements.	-68
Transfer the tank supports to the Propulsion System	-62
Decrease the antenna support structure per revised estimate	- 5
Decrease the forward bulkhead per revised estimate	- 3
Decrease the aft bulkhead per revised estimate	-30
Add tank support shelf in the equipment bay sextant I	+30
Delete insulation from the following items	-432
Outer shell -273 Fairing - Command to Service -10 Aft Bulkhead -149	
Add the insulation for complete service module structure	+432
Decrease the insulation using lower density material	-198
ELECTRONICS	(-6)
Decrease Instrumentation sensors consistent with present requirement	-10
Add instrumentation signal conditioning	+15
Delete IFTS panel installation	-11
REACTION CONTROL SYSTEM	(-50)
Reduce component unit weights consistent with specifications and revised schematics	-25
Reduce structural provisions for modular panels per revised estimate. Panels are partially supported by basic structure joints, and radial beams.	- 25
15	- CONFID



SERVICE MODULE

CURRENT WEIGHT EMPTY CHANGES

ELECTRICAL POWER SYSTEM	(+146)
Increase fuel cell power pack consistent with specification	+87
Increase fuel cell hydrogen and oxygen system per Beech Status	+102
Add wiring installation for supercritical gas system	+21
Delete fuel cell module mounts as they are included with the fuel cell power pack	-32
Reduce plumbing and valves per revised estimate	-32
ENVIRONMENTAL CONTROL SYSTEM	(- 28)
Reduce water-glycol consistent with present requirements	-8
Reduce water-glycol components per airesearch status	-2
Reduce water-glycol plumbing and hardware per revised estimate	-14
Reduce water supply plumbing and hardware per revised estimate	-4
PROPULSION SYSTEM	(-9)
Reduce component unit weights to agree with specification	-9
TOTAL SERVICE MODULE WEIGHT EMPTY CHANGES	-591





SERVICE MODULE

CURRENT USEFUL LOAD CHANGES

Change supercritical reactants per	revised estimate:	+20
Usable reactants	(-42)	
Hydrogen	-4	
Oxygen	-3 8	
Unusable Reactants	(+62)	
Hydrogen	+8	
Oxygen	+54	
Change environmental control fluids	per revised analysis	+11
Reduce back pack oxygen	- 9	
Add provisions for emergency cond	i. +17	
Reduce residuals	-2	
Increase oxygen reserve	+5	
TOTAL SERVICE MODULE USEFUL LOAD CHA	ANGES	+31



LAUNCH ESCAPE SYSTEM

WEIGHT STATUS

	PREVIOUS	CHANGE	CURRENT	EASIS FOR CURRE			OR CURRENT	
ITEM	STATUS 11-1-62	TO CURRENT	WEIGHT 12-1-52	%EST	%CALC	%ACT		
Structure	1091	+3	1094	11	89			
Electrical System	20		20	100				
Propulsion System Main Thrust Jettison Pitch Control	4826 440 55		4826 440 55	50 1 75	50 99 25			
LES - NO BALLAST	6432	+3	6435	40	60			
BALLAST	63	+157	220	100	·			
TOTAL L.E.S.	6495	+160	6655	42	58			



LAUNCH ESCAPE SYSTEM

CURRENT WEIGHT CHANGES

ITEM	CURRENT WEIGHT 12-1-62
LAUNCH ESCAPE SYSTEM	
Structure Increase or decrease, as applicable, the weight of the following items based on calculations of released drawings versus previous estimates:	(+3) +3
Tower Flow Separator and Skirt Jettison Motor Skirt Pitch Motor Structure Pitch Motor Structure Flow Separator Flow Separator Insulation Flow Separator Insulation -9 -9 -9 -9 -9 -9 -9 -14 -15	
Ballast Increase ballast consistent with current balance requirement	(+15 7) +15 7
TOTAL LAUNCH ESCAPE SYSTEM	+160



ADAPTER

WEIGHT STATUS

	PREVIOUS	CHANGE	CURRENT	BASIS	FOR CURRENT	
ITEM	STATUS TO 11-1-62 CURRENT	_	WEIGHT 12-1-62	%EST	%CAL	%ACT
Structure	2892		2892			
Electrical	70		70			
Separation System	142		142			
Posigrade Rocket (4)	156		156			
TOTAL ADAPTER	3260		3260	100		



WEIGHT HISTORY COMMENTS

LAUNCH ESCAPE SYSTEM

The target weight established for the LES is 6,300 pounds, excluding ballast. This weight was based on the September status weight of 6,600 pounds including the necessary ballast to provide currently determined aerodynamic stability to prevent tumbling.

The original target weight of 5,900 pounds as reported in the June Status, SID 62-99-5, was based on an attitude controlled configuration. The current configuration weight includes a flow separator, pitch motor, and ballast not included in the original target weight.

COMMAND MODULE

The target weight established for the Command Module is 8,500 pounds. an estimated weight breakdown for the target weight is provided for comparative purposes.

The original target weight of 8.340 pounds as reported in the June Status, SID 62-99-5, did not include the proposed increases nor the category I reductions presented in the July briefing and incorporated in the July Status report

SERVICE MODULE

The target weight established for the Service Module less usable propellant is 11,000 pounds. An estimated weight breakdown for the target weight is provided for comparative purposes. This configuration is sized for 45,000 pounds usable propellant for the 25,000 pound LEM.

The original target weight of 8,675 for the burnout condition was based on a lunar landing configuration sized for 31,000 pounds usable propellant...



WEIGHT HISTORY COMMAND MODULE

	ORIGINAL TARGET WT.	TARCET WEIGHT	AUTHORIZED CHANGES	AUTHORIZED WEIGHT 12-1-62
Structure	3670	3720	+399	41 1 9
Crew Systems	565	6 90	+2	692
Communication & Instrumentation	944	785		785
Guidance & Navigation	3 10	310	+64	374
Stabilization & Control	175	195		195
Reaction Control	183	195		195
Electrical Power	354	390		390
Environmental Control	228	255		255
Earth Landing	530	610	-106	504
WEIGHT EMPTY	6959	7150	+359	7509
Crew	528	528		528
Suits & Personal Equipment	82	126		126
Survival Water	54	18		18
Food & Containers	90	90		90
Reaction Control Propellant	210	210		210
Environmental Control Fluids	167	128		128
Scientific Payload	250	250		250
GROSS WEIGHT	8340	8500	+ 3 59	8859



- CONFIDENTIAL -

COMMAND MODULE WEIGHT HISTORY

WEIGHT EMPTY AUTHORIZED CHANGES

STRUCTURE		(+399)
The following authorized structural weight changes were reported in the 1 November 1962 Weight and Balance Report:		+381
Approval for use of the low density AVCO ablator in the heat shield system was withheld pending the accumulation of more design data.	+380	
Delete readily removable items from the impact attenuation system & redesign as required.	+1	
The 1 November 1962 Weight and Balance report removed 149 pounds for impact attenuation fittings. However, 18 pounds has been found to be not readily removable with the current design concept of not removing provisions for impact attenuation that are buried in the honeycomb panels and welded or bonded to the face sheets.	+18	
COMMUNICATION & INSTRUMENTATION		(+2)
Add a loudspeaker in the crew compartment per NASA request.		+2
GUIDANCE & NAVIGATION		(+64)
Increase the Guidance and Navigation per Recent Weight report from M.I.T. Since NAA does not have weight control responsibility for the M.I.T. Design, the weight changes in their Weight and Balance report will be considered as authorized changes.		+64
EARTH LANDING		(-106)
The removal of the impact attenuation system per TWX SM 032 dated 23 July 1960 was reported in the 1 November 1962 Weight and Balance report.		-106
TOTAL COMMAND MODULE WEIGHT EMPTY CHANGES	-	+359



WEIGHT HISTORY

SERVICE MODULE

	ORIGINAL TARGET WT.	TARGET WEIGHT	AUTHORIZED CHANGES	AUTHORIZED WEIGHT 12-1-62
Structure	2810	3203		3203
Electronics	216	145		145
Reaction Control	254	737		73 7
Electrical Power	1076	1203		1203
Environmental Control	413	250		250
Propulsion System Engine Installation	3 75	606		606
Propellant System	1928	2456		2456
WEIGHT EMPTY	7072	8600		8600
Usable RCS Propellant Usable Fuel Cell Reactants Environ. Control Fluids Main Propulsion Helium Main Prop. Residuals Unusable RCS Propellant Unusable Fuel Cell Reactants	400 380 288 97 300 20 28	611 479 193 139 900 61 17		611 479 193 139 900 61 17
BURNOUT WEIGHT	8595	11000		11000
Main Propellant	31000	45000		45000
CROSS WEIGHT	39595	56000		56000



POTENTIAL WEIGHT AND C.G. CHANGES

COMMAND MODULE

STRUCTURE			(-604)
Decrease ablative to agree with the recommended by Avco. The change is the new design is as follows:			- 384
	OLD DESIGN	NEW DESIGN	
Type of wet blend ablative Lift-off temperature Blunt face prior to re-entry Back face after touchdown Density (lb./ft.3)	5026–22 300°F 300°F 600°F 58	5026-39 125°F 100°F 600°F or gre 37	ater
Decrease primary structure based on The present structure has several reduction in addition to reduction A summary of these potential change Reduce skin gauges, weld lands, of to minimum structural requirement to optimize disign as applicable	areas of post from pot ges is as force densitents and im	otential weight ential criteria ollows: ies and fitings	changes.
Remove complete provisions for the System that have not been previous removable items. This consist in the existing design.	iously dele	ted as readily	-170
STABILIZATION AND CONTROL			(+25)
Increase M-H electronic control ampl of signal ground isolator	lifier for	incorporation	+25
TOTAL POTENTIAL WEIGHT CHANGES - COMMAN	D MODULE		- 659

A potential weight increase in the Command Module due to LFM docking will be included at a later date.



POTENTIAL WEIGHT AND C.G. CHANGES

SERVICE MODULE

STRUCTURE	(-114)
The removal of insulation from the radial beams may result from current Thermodynamics analysis.	-64
Revision of the Separation System to an explosive Bolt System	- 50
TOTAL POTENTIAL WEIGHT CHANGES - SERVICE MODULE	-114



DETAIL WEIGHT STATEMENT

COMMAND MODULE

SUMMARY

ITEM		CURRENT WEIGHT 12-1-62
WEIGHT EMPTY		7960
Structure	4482	
Crew Systems	513	
Communication & Instrumentation	88 9	
Guidance & Navigation	431	
Stabilization & Control	214	
Reaction Control	26 7	
Electrical Power	429	
Environmental Control	262	
Earth Landing	493	
USEFUL LOAD		1370
Crew Systems	770	
Reaction Control	210	
Environmental Control	140	
Scientific Payload	250	
GROSS WEIGHT		93 5 0



DETAIL WEIGHT STATEMENT

COMMAND MODULE

STRUCTURE

ITEM	CURRENT WEIGHT 12-1-62
STRUCTURE	
Heat Shield Forward Compartment Crew Compartment Aft Compartment	(1489) 236 628 625
Inner Structure Forward Section Aft Sidewall Aft Bulkhead	(1060) 523 359 178
Secondary Structure Internal Cahin Service Module Attach	(2 6 3) 253 10
Ablation Material	(1484)
Microfiber Insulation	(186)
TOTAL STRUCTURE	4482 1489



DETAIL WEIGHT STATEMENT

COMMAND MODULE

CREW SYSTEMS

ITEM	CURRENT WEIGHT 12-1-62
CREW SYSTEMS	
Personal Radiation Dosimeter (NASA) Portable Life Support System (2) (NASA) Bio-Medical Inst. (NASA) Personnel Communications (NASA) Seat Liners & Restraint Harness Sleeping Restraints Curtain, Sleeping & Relief Privacy Waste Management Lighting System Garments - Constant Wear (NASA) Emergency Water Device Survival Kit - Collective (1) (NASA) Shoe Straps Food Reconstitution Device Log Book, Pencils etc. Hatch Egress Lap Board (2) Manual - Maint. Maps, & Case Suit Umbilical Hose In-Flight Test Maintenance Tool Belt Structural Seats & Supports Nuclear Radiation Detectors In-Flight Maintenance Tool Food Preperation Table	5 60 0 5 36 4 0 25 15 9 2 54 2 4 1 3 2 6 10 1 25 8 7 1
TOTAL CREW SYSTEMS	513



DETAIL WEIGHT STATEMENT COMMAND MODULE COMMUNICATIONS & INSTRUMENTATION

ITEM	CURRENT WEIGHT
	12-1-62
TELECOMMUNICATIONS	
Lower Bay	(265)
C-Band Transponder	16
DSIF TransRec.	16
DSIF Power Amplifier	13
DSIF Spare Module	18
VHF FM Transmitter	10
VHF AM TransRec.	14
HF Transceiver	10
VHF Recovery Beacon	7
Multiplexer	9
Antenna Switches	4
Spares	19
PCM Telemetry	_,
Analog Commutator	18
Analog Commutator	9
A/D Converter	7
Programmer	15
Calibrator	5
Power Supply	5 7
Signal Conditioner	10
Input Analog Patch	3
Digital Patch Panel	4
Output Analog Patch	3 4 3 15
Recorder	15
Audio Center	5
Premodulation Processor	18
Central Timing Equipment	10
Remote Equipment	(12 2)
VHF/2-KMC OMNI Antenna	53
Antennas & Transmission	30
TV Camera	4
Instrumentation Sensors	35
Supports	(12)
Electrical Provisions	(95)
Electronic Interface Provisions	(8)
Cooling Provisions	(20)
TOTAL TELECOMMUNICATION (to be brought forward)	522



DETAIL WEIGH	T STATEMENT	
ONFIDENTIAL COMMAND		CURRENT
COMMUNICATION AND	INSTRUMENTATION	WEIGHT
ITEM		12-1-62
CONTROLS AND DISPLAYS		
Main Display Panel Control Station		(62)
Integrated Display	C&I	10
GMT Clock	C&I	1
Computer Data Insert & Display	G&N	15
Time to & from Clock	G&N	
SCS Control Panel	SCS	2 5 3
Delta Velocity Control	SCS	3
Flight Director Attitude Indicator	SCS	11
Gimbal Angle Indicator	SCS	2
Altimeter	SCS	2 2 8
Survival Indicator	E&A	
Master Caution Lights	Ŭ	3
Main Display Panel Center Station		(37)
Audio Panel	C&I	1
8 Day Clock	C&I	ī
Indicator Light & Abort Light	Ü	ī
Reaction Control System	Ü	7
Service Module Propulsion	Ŭ	7
Booster Situation Indicator	U	2
ECS - Gas Control	U	6
ECS - Liquid Control	U	6
Service Module Quad. Temp. Ind.		3 2
SCS Power Cont.		2
IFTS Scan Select		1
Main Display Panel System Management S	Station	(40)
Communications Control Panel	C&I	8
Antenna Control	C&I	3
Abort Light & Master Caution	n Ogra	3
Power Distribution	Ū	12
Fuel Cell Reactants	Ū	9
Cryogenic Storage	Ū	ź
Main Dissiles Description Description		(05)
Main Display Panel Installation Provis	ions	(25)
Main Display Right Hand Console	U	(11)
Nuclear Detection Display		3
Installation Provisions		3 6
Lighting Cont.		1
Audio Panel		1
Main Display Left Hand Console		(12)
Installation Provisions		7
Lighting Cont.		1
Audio Panel		1
Boost Emergency, Earth Landing		3
Electrical Provisions		(29)
Environmental Provisions		(9)
Loudspeaker		(2)

TOTAL CONTROLS AND DISPLAYS (to be brought forward)



<u>_CONFIDENTIAL</u>

DETAIL WEIGHT STATEMENT

COMMAND MODULE

COMMUNICATION AND INSTRUMENTATION

ITEM	CURRENT WEIGHT 12- 1-62
IN-FLIGHT TEST (RIGHT BAY FORWARD) Crew Readout Panel Manual Test Unit Comparator Programmer Stimuli Generator Panel Assy Installation Provisions & Connectors In-Flight Test - GSE Electrical Provisions	(125) 3 15 12 11 24 15 5 40
CREW AREA CONTROLS Manual Control - Three Axis Manual Control - Translation & Thrust	(15) 7 8
TOTAL IN-FLIGHT TEST & CREW AREA CONTROLS	140
TOTAL CONTROLS AND DISPLAYS	227
TOTAL TELECOMMUNICATION	522
TOTAL COMMUNICATIONS AND INSTRUMENTATION	889



DETAIL WEIGHT STATEMENT

COMMAND MODULE

GUIDANCE & NAVIGATION

ITEM	CURRENT WEIGHT 12-1-62
GUIDANCE & NAVIGATION	
Lower Equipment Bay Inertial Platform Sextant	60 12
Telescope - Scanning Map & Visual Display	9
Display & Control - Navigation Display & Control - Computer	40 15
Navigation Base Computer	45 100
Power Servo Assy Coupling Display Unit	29 6
Junction Box Cabling	11 40
Spares NAA Cabling	40 15
TOTAL GUIDANCE & NAVIGATION	431



DETAIL WEIGHT STATEMENT

COMMAND MODULE

STABILIZATION AND CONTROL

ITEM	CURRENT WEIGHT 12-1-62
STABILIZATION AND CONTROL	
Rate Gyro Package Body Mounted Gyro Package Electronic Control Package No. 1 Electronic Control Package No. 2 Electronic Control Package No. 3 Accelerometer Display Electronics Spare Gyro - BMAG (2) Spare Gyro - Rate Spare Plug-in Module BMAG Coupler	(159) 5 15 30 30 31 5 15 2 1 12 13
Supports	(12)
Electrical Provisions	(16)
Environmental Control Provisions	(27)
TOTAL STABILIZATION AND CONTROL	214



DETAIL WEIGHT STATEMENT

COMMAND MODULE

REACTION CONTROL SYSTEM

ITEM	CURRENT WEIGHT 12-1-62
REACTION CONTROL SYSTEM	
Propellant System Tanks Expulsion Devices & Tank Supports Plumbing, Fittings & Insulation Valves & Regulators Sensors Supports - Plumbing & Equipment	(81) 17 12 23 21 1
Pressure System Tanks (4500 psi) Tank Supports Plumbing, Fittings & Insulation Valves & Regulators Sensors Helium Supports - Plumbing & Equipment	(49) 9 3 4 24 2 1 6
Engine System Engines Supports	(114) 108 6
Electrical Provisions	(23)
TOTAL REACTION CONTROL SYSTEM	<u> 267</u>



DETAIL WEIGHT STATEMENT

COMMAND MODULE

ELECTRICAL POWER

ITEM	CURRENT WEIGHT 12-1-62
ELECTRICAL POWER	
Energy Source	(60.0)
Battery - Main (2)	36.0
Battery - Recovery (1) Installation Provisions	18.0
Installation Frovisions	6.0
Power Conversion	(107.6)
Inverter (3)	90.0
Inverter Controls	3.0
Battery Charger & Controls Installation Provisions	5.0
installation Frovisions	9.6
Power Distribution & Control	(162.0)
Power Distribution Equipment	
Circuit Breakers	6.0
Battery Controls	5.0
No. 1 and No. 2 AC Bus Control DC Busses (Diodes, etc.)	15.0
AC Busses	10.0
Utility System Controls	5.0 15.0
Mounting Hardware	2.0
Sequencer	10.0
Right Hand Circuit Breaker Panel	13.0
Terminal Panels	5.0
Power Distribution Wiring & Provisions	40.0
Lighting Wiring & Provisions	5.0
Ground Power Provisions	6.0
Power Control Panel Connectors	3.0
Installation Provisions	22.0
Electrical - Common Utility	(99•4)
Utility Wiring and Circuit Components	20.0
Left Hand Circuit Breaker Panel	7.0
Umbilicals	47.0
Posigrade Adapter Separation System	10.0
Launch Escape System Separation	5.0
Service Module Electrical Initiation	5.0
Installation Provisions	5.4

TOTAL ELECTRICAL POWER

429.0



DETAIL WEIGHT STATEMENT

COMMAND MODULE

ENVIRONMENTAL CONTROL SYSTEM

ITEM	CURRENT WEIGHT 12-1-62
ENVIRONMENTAL CONTROL SYSTEM	
Pressure Suit Circuit Subcontractor Components Ducting, Conn., Clamps, etc.,	(71.0) 58.9 12.1
Water-Glycol Circuit Subcontractor Components Water-Glycol Plumbing, etc.	(49.4) 31.5 14.3 3.6
Pressure & Temp. Control Subcontractor Components Ducting	(16.5) 15.8 0.7
Oxygen Supply System Subcontractor Components Plumbing	(17.0) 14.0 3.0
Water Supply System Subcontractor Components Plumbing	(27.8) 23.0 4.8
Subcontractor Common Items Brackets, Plumbing, Elect. Wiring Instrumentation	(32.6) 20.0 12.6
Gas Analyzer	(16.7)
Supports	(10.0)
Electrical Provisions	(21.0)
TOTAL ENVIRONMENTAL CONTROL SYSTEM	262.0



DETAIL WEIGHT STATEMENT

COMMAND MODULE

EARTH LANDING SYSTEM

ITEM	CURRENT WEIGHT 12-1-62
EARTH LANDING SYSTEM	
Parachute System Drogue Main Cluster Sequence Control Location Aids Attach Provisions	(46 3) 40 385 22 11 5
Forward Heat Shield Release System	25
Electrical Pyrotechnic Initiation Provisions	5
TOTAL EARTH LANDING SYSTEM	493



DETAIL WEIGHT STATEMENT

COMMAND MODULE

USEFUL LOAD

ITEM	CURRENT WEIGHT 12-1-62
CREW SYSTEMS Crew (3) (50, 70, 90, Percentile) Pressure Garment Assy (3) (NASA) Food Food Containers Water - Survival Personal Hygiene Equipment Biomedical Instrumentation (NASA) Medical Equipment Chemical Disinfectant	(770) 528 90 75 15 18 23 2 15
REACTION CONTROL RCS Propellant	(210) 210
ENVIRONMENTAL CONTROL Lithium Hydroxide Activated Charcoal Containers for LiOH & Charcoal Oxygen - Re-entry Water-Launch & Re-entry Cooling Freon	(140) 112 4 6 2 10 6
SCIENTIFIC PAYLOAD	(250)
TOTAL COMMAND MODULE USEFUL LOAD	1370



DETAIL WEIGHT STATEMENT

SERVICE MODULE

SUMMARY

ITEM	CURRENT WEIGHT 12-1-62
WEIGHT EMPTY	7793
Structure	2673
Electronics	166
Reaction Control	591
Electrical Power	1243
Environmental Control	58
Propulsion	3062
USEFUL LOAD	2412
Reaction Control	672
Electrical Power	499
Environmental Control	202
Propulsion	1039
BURNOUT WEIGHT	10205
MAIN PROPELLANT	45000
GROSS WEIGHT	55205



DETAIL WEIGHT STATEMENT

SERVICE MODULE

STRUCTURE

ITEM	CURRENT WEIGHT 12-1-62
STRUCTURE	
Basic & Secondary Structure Radial Beams Internal Structure & Engine Compartment Closeout Outer Shell Fairing - Command to Service Engine Support Tank Supports Antenna Support Structure Forward Bulkhead Including Ring Aft Bulkhead Separation Provisions Tank Support Shelf Insulation	477 45 920 200 41 0 30 148 478 70 30 234
TOTAL STRUCTURE	2673
	: 1.12



DETAIL WEIGHT STATEMENT

SERVICE MODULE

ELECTRONIC SUBSYSTEMS

ITEM	CURRENT WEIGHT 12-1-62
ELECTRONICS SUBSYSTEM	
Communications Antenna Dish Antenna Gimbals Antenna Deployment Booms Antenna Coax Cabling Antenna Coax Cabling Supports Antenna Control Electrical Provisions	(72) 10 13 25 16 3
Instrumentation Sensors Electrical Provisions Supports Signal Conditioner	(64) 30 14 5
In-Flight Test Provisions In-Flight Test & GSE Electrical Provisions	(30) 30
TOTAL ELECTRONICS SUBSYSTEMS	166



DETAIL WEIGHT STATEMENT

SERVICE MODULE

REACTION CONTROL

ITEM	CURRENT WEIGHT 12-1-62
REACTION CONTROL SYSTEM	
Propellant System Tanks Expulsion Devices & Tank Supports Plumbing, Fittings & Insulation Valves & Regulators Sensors Supports - Plumbing & Equipment	(153) 46 36 17 32 6 16
Pressure System Tanks (4500 psi) Tank Supports Plumbing, Fittings & Insulation Valves & Regulators Sensors Helium Supports - Plumbing & Equipment	(132) 22 6 4 74 9 3 14
Engine System Engines Reflectors & Insulation	(165) 55 110
Structural Provisions	(80)
Electrical Provisions	(61)
TOTAL REACTION CONTROL SYSTEM	591



DETAIL WEIGHT STATEMENT

SERVICE MODULE

ELECTRICAL POWER

ITEM	CURRENT WEIGHT 12-1-62
ELECTRICAL POWER	
Fuel Cell Power System	(1110.0)
Fuel Cell Power Pack	732.0
Intermodular - Radiator Plumbing	12.0
Fuel Cell Module Mount Attach.	2.0
Fuel Cell H2 System	
Subcontractor Components	122.0
Plumbing and Valves	3.0
Fuel Cell and ECS O2 System	_
Subcontractor Components	207.0
Plumbing and Valves	4.0
Water Glycol - Fuel Cell Heat Transfer System	7.0
Elect Wiring - Supercritical Gas	21.0
Power Distribution	(101.0)
Relays	10.0
Power Switch	5.4
Motor Switch	1.5
Umbilicals	64.1
Wiring & Busses	15.0
Supports	5.0
Electrical Utilities	(32.0)
Command - Service Separation System	3.0
Posigrade Adapter Separation System	10.0
Electrical Initiation of Pyrotechnics	17.0
Supports	2.0
TOTAL ELECTRICAL POWER	1243.0



DETAIL WEIGHT STATEMENT

SERVICE MODULE

ENVIRONMENTAL CONTROL SYSTEM

I TEM	CURRENT WEIGHT 12-1-62
ENVIRONMENTAL CONTROL SYSTEM	
Water-Glycol Circuit - Airesearch Subcontractor Components Plumbing and Hardware Radiator Provisions Water - Glycol Supports	(23.6) 4.9 4.4 5.6 4.0 4.7
Water Supply System Subcontractor Components Plumbing and Hardware Supports	(8.4) 0.4 6.0 2.0
Oxygen Supply System Plumbing and Supports	(3.0) 3.0
Electrical Provision	(23.0)
TOTAL ENVIRONMENTAL CONTROL SYSTEM	58.0



DETAIL WEIGHT STATEMENT

SERVICE MODULE

MAIN PROPULSION

ITEM	CURRENT WEIGHT 12-1-62
MAIN PROPULSION	
Propellant System	(1 489)
Tanks & Integral Doors	990
Tank Skirts	20 0
Plumbing, Fittings & Insulation	129
Valves	9
Quantity Indication	7 0
Mixture Ratio Control	12
Supports - Plumbing & Equipment	79
Pressure System	(941)
Tanks (4500 psi)	800
Tank Supports	30
Plumbing, Fittings & Insulation	24
Valves, Regulators & Heat Exchanger	49
Supports - Plumbing & Equipment	38
Engine System	(606)
Engine	606
Electrical Provisions	(26)
TOTAL MAIN PROPULSION SYSTEM	3062



DETAIL WEIGHT STATEMENT

SERVICE MODULE

USEFUL LOAD

ITEM	CURRE WEIGH 12-1-	T
REACTION CONTROL	(67	'2)
Usable Reaction Control System Propellant	61	
Unusable Reaction Control System Propellant	6	1
ELECTRICAL POWER	(49	9)
Usable Superciritical Reactants		
Hydrogen (incl. 10% reserve)		8
Oxygen (incl. 10% reserve) Unusable Supercritical Reactants	38	,3
Hydrogen		9
Oxygen	5	9
ENVIRONMENT CONTROL	(20	12)
Oxygen - ECS	20	•
PROPULSION	(103	10)
Main Propulsion Helium	13	
Main Propellant Residuals	90	-
Trapped	225	
Mixture Ratio Tolerance	150	
MIXCUIS RACIO TOTELANGS	450	
Loading Tolerance	225	
TOTAL USEFUL LOAD	24]	12



DETAIL WEIGHT STATEMENT

LAUNCH ESCAPE SYSTEM

SUMMARY

ITEM	CURRENT WEIGHT 12-1-62
LAUNCH ESCAPE SYSTEM	
Structure	(1094)
Tower Assy.	306
Flow Separator and Skirt	285
Jettison Motor Skirt	94
Pitch Motor Structure	157
Nose Cone and Ballast Support	111
Attaching Parts	25
Tower Insulation	45
Skirt Insulation	26
Flow Separator Insulation	45
Ballast	(220)
Propulsion	(5321)
Escape Motor	4826
Jettison Motor	440
Pitch Control Motor	55
Electrical Power	(20)
TOTAL LAUNCH ESCAPE SYSTEM	6655



- GONFIDENTIAL

DETAIL WEIGHT STATEMENT

ADAPTER

SUMMARY

ITEM	CURRENT WEIGHT 12-1-62
ADAPTER	
Structure Panels Frames Thermal Insulation	(2892) 1915 421 556
Electrical Power	(70)
Separation System	(142)
Posigrade Rockets (4)	(156)
TOTAL ADAPTER	3260